

United States of America
Department of Transportation -- Federal Aviation Administration
Supplemental Type Certificate

Number SR09318RC

This certificate issued to

Global Helicopter Technology, Inc.
4846 S. Collins Street
Arlington, TX 76018-1108

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 7 of the Civil Air Regulations.

Original Product -- Type Certificate Number : H1SW

Make : Bell

Model : 204B

Description of Type Design Change: .

Fabrication, modification, and installation of a new vertical fin spar in accordance with Global Helicopter Technology, Inc., Drawing number VTF-030-800, Revision, E, dated June 6, 2001, or later FAA approved revision.

Limitations and Conditions:

(See continuation sheet 3 of 3)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application : March 27, 1901

Date reissued : January 27, 2004

Date of issuance : June 06, 2001

Date amended :



By direction of the Administrator

(Signature)

Carl F. Mittag, Manager
Rotorcraft Certification Office,
Southwest Region

(Title)

United States of America
Department of Transportation -- Federal Aviation Administration
Supplemental Type Certificate
(Continuation Sheet)

United States of America
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Supplemental Type Certificate
(Continuation Sheet)

Number SR09318RC

Date of Issuance: June 06, 2001

Date Amended: June 06, 2001

Reissuance Date: January 27, 2004

Limitations and Conditions (Continued):

Approval of this change in type design applies only to rotorcraft models specified above, that are equipped with a standard tail boom with part number 205-030-899-109, fin assembly installed. This approval should not be extended to any other rotorcraft of this type, on which other previously approved modifications are incorporated, unless it is determined that the relationship between this change and any other previously approved modifications, including changes in type design, will introduce no adverse effect upon the airworthiness of the helicopter. This modification must be maintained in accordance with Global Helicopter Technology, Inc., Instructions for Continued Airworthiness Report Number GHT-445-204B, Revision O, dated March 26, 2001, or later FAA accepted revision. This modification is not eligible for aircraft equipped with Bell Helicopter 212 Tail Rotor or other Tail Rotor Retrofit Systems installed (212-011-701, Hub and Blade), or with alternate horsepower engines installed that are not specified on the Type Certificate Data Sheet. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

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